

APPROVED FOR PUBLIC RELEASE DISTRIBUTION UNLIMITED

REPORT NO. 89-R-04 AFPEA PROJECT NO. 88-P-103



Larry M. Nugent

Mechanical Engineer

AUTOVON 787-3734

Commercial (513) 257-3734

QUALIFICATION TESTING OF THE CNU-459/E KU-BAND ANTENNA CONTAINER

HQ AFLC/DSTZ
AIR FORCE PACKAGING EVALUATION ACTIVITY
Wright-Patterson AFB OH 45433-5999

June 1989

NOTICE

When government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related government procurement operation, the United States Government thereby incurs no responsibility whatsoever, and the fact that the government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner ficensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto. This report is not to be used in whole or in part for advertising or sales purposes.

ABSTRACT

Aeronautical Systems Division, ASD/VXAL, requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to design, prototype, and qualify a container for the ku-band antenna used on Combat Talon II aircraft.

The CNU-459/E prototype container was designed, fabricated and tested at the AFPEA, HQ AFLC/DSTZ, Wright-Patterson AFB, OH 45433-5999. The container is environmentally sealed with a humidity indicator, desiccant port, and pressure relief valve. The container base has wheels and the cover doubles as a transport stand to eliminate the need for ground support equipment. The container is designed to protect one ku-band antenna during worldwide shipment, storage, and handling.

The container test plan was derived from Military Specification MIL-C-5584D. The tests were conducted in accordance with Federal Test Method Standard No. 101, Military Standard 648, Military Standard 1489, and Military Specification MIL-C-5584D.

Results of the tests conducted on the prototype container are acceptable. The container design will adequately protect one ku-band antenna during worldwide shipment and storage.

PREPARED BY:

LARRY M. NÜGENT () Mechanical Engineer

AF Packaging Evaluation Activity

REVIEWED BY:

TED HINDS Chief, Design Branch

AF Packaging Evaluation Activity

PUBLICATION DATE:

27 JUN 1989

APPROVED BY

CHARLIE P. EDMONSON

Chief,

AF Packaging Evaluation Activity

1. La- 11

TABLE OF CONTENTS

	<u>Page</u>
ABSTRACT	i
TABLE OF CONTENTS	ii
INTRODUCTION	
BACKGROUND	1
PURPOSE	1
TEST SPECIMEN	1
TEST OUTLINE AND TEST EQUIPMENT	2
TEST PROCEDURES AND RESULTS	2
TEST NO. 1, WEIGHT TEST	2
TEST NO. 2, LEAK TEST	3
TEST NO. 3A, CORNERWISE-DROP (ROTATIONAL) (+140°F) TEST	3
TEST NO. 3B, EDGEWISE-DROP (ROTATIONAL) (+140°F) TEST	3
TEST NO. 3C, PENDULUM-IMPACT (+140°F) TEST	3
TEST NO. 4, LEAK TEST	3
TEST NO. 5A, CORNERWISE-DROP (ROTATIONAL) (-20°F) TEST	4
TEST NO. 5B, EDGEWISE-DROP (ROTATIONAL) (-20°F) TEST	4
TEST NO. 5C, PENDULUM-IMPACT (-20°F) TEST	4
TEST NO. 6, LEAK TEST	4
TEST NO. 7, VIBRATION FATIGUE TEST	4
TEST NO. 8, LEAK TEST	5
TEST NO. 9, REPETITIVE SHOCK (SUPERIMPOSED LOAD) TEST	5 1
TEST NO. 10, LEAK TEST	5 0
TEST NO. 11, SUPERIMPOSED LOAD TEST	6
The state of the s	

COPY

Availability Codes

Dist Avail and for Special

TEST NO. 12, LEAK TEST	•
TEST NO. 13, STAND OFF TEST	•
TEST NO. 14, MECHANICAL HANDLING TEST	•
TEST NO. 15, HOISTING STRENGTH TEST	7
TEST NO. 16, LEAK TEST	7
TEST NO. 17, COVER HANDLE PULL TEST	8
TEST NO. 18, PUNCTURE RESISTANCE TEST	8
TEST NO. 19, LEAK TEST	8
TEST NO. 20, TOWING TEST	8
TEST NO. 21, STRUCTURAL INTEGRITY TEST	9
CONCLUSIONS	9
RECOMMENDATIONS	g
TABLE 1, CONTAINER TEST PLAN	10
FIGURE 1, CNU-459/E CORNER NUMBERING	21
FIGURE 2, CNU-459/E PROTOTYPE CONTAINER	22
FIGURE 3, CNU-459/E YOKE ASSEMBLY ATTACHMENT	22
FIGURE 4, CNU-459/E COVER WHEEL STORAGE	23
FIGURE 5, CNU-459/E YOKE FRAME	23
FIGURE 6, CNU-459/E TEST LOAD	24
FIGURE 7, HIGH TEMPERATURE PENDULUM-IMPACT TEST	24
FIGURE 8, FORKLIFT HANDLING TEST	25
FIGURE 9, FOUR RING HOISTING TEST	25
FIGURE 10, COVER HANDLE PULL TEST	26
DISTRIBUTION LIST	27

APPENDICES:

APPENDIX 1, DÉTAILED ACCELERATION RESULTS

INTRODUCTION

BACKGROUND: Aeronautical Systems Division (ASD/VXAL), Wright-Patterson AFB OH 45433-5000 requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to design, prototype, and perform qualification testing on a ku-band antenna container. The CNU-459/E prototype was designed and fabricated at the AFPEA, HQ AFLC/DSTZ, Wright-Patterson AFB, OH 45433-5999.

<u>PURPOSE</u>: The purpose of this project was to determine if the CNU-459/E container design will protect the contents, one ku-band antenna for Combat Talon II aircraft, during worldwide shipment, storage, and handling.

TEST SPECIMEN

The CNU-459/E prototype was subjected to extensive testing. The corners of the container were numbered counterclockwise from the left aft corner as shown in figure 1.

The CNU-459/E prototype is a controlled-breathing container with a pressure relief valve, humidity indicator, and desiccant port (see figure 2). The container is designed to limit the transmission of shocks to the ku-band antenna to 25 Gs when subjected to the conditions in Military Specification MIL-C-5584D. Twelve wide-handle latches are designed to allow quick access to the container contents without the use of tools. The antenna is attached to a yoke in the container by five bolts. The yoke is attached to the cradle by four wide-handle latches (see figure 3). The antenna can rest on the cradle without the yoke during aircraft installation and removal. Wheels on the base may be lowered to allow the container to be towed or pushed with minimal effort. Cover wheels and resting posts stowed in the container base may be attached to the container cover so it may be similarly used to transport or hold one antenna (see figure 4).

Construction: The container consists of aluminum extrusions for the base walls, the lower cover walls, the wall stiffeners, the stacking interface, the skid base, the yoke, and the cradle. Sheet aluminum is used for the upper cover walls and the top and bottom surfaces. Rubber pads on the cradle and cover resting posts prevent damage to the antenna while resting there. PPP-C-1752, Type II, Class 1, (two pound density polyethylene foam) provides cushioning between the cradle assembly and the floor, and the yoke assembly and the cover of the container. A silicone gasket provides a seal between the container base and the container cover.

The CNU-459/E prototype container tested differs from the proposed production container in the following aspects:

- a. Sixteen latches (four per side) will be used instead of twelve (three per side) to secure the cover to base.
- b. A frame has been added around the yoke to prevent bumping the cover into the antenna when the cover is being removed from the container (see figure 5).
- c. The desiccant port and pressure relief valve have been moved from the cover to the base.
- d. Towbar attachments and cover lifting rings were added and tested after all other testing was complete.
- e. Cover and base have been rotated 90° relative to the antenna, the yoke and the cradle.
- f. The cushions may not be glued sections; must be either one solid piece or heat bonded.

TEST OUTLINE AND TEST EQUIPMENT

Test Plan: Tests were conducted in accordance with AFPEA Test Plan 88-P-103 (see table 1). The tests used were selected to meet the qualification requirements in Military Specification MIL-C-5584D. Test methods and procedures used were as outlined in Federal Test Method Standard No. 101 (FED-STD-101), Military Standard 648 (MIL-STD-648), Military Standard 1489 (MIL-STD-1489), and Military Specification MIL-C-5584D. Any modifications to the standard procedures are noted in the test plan or the results.

Test Loads: All tests were conducted using the ku-band antenna test load fabricated at the AFPEA (see figure 6). The test load weighs 206 pounds and simulates the center of gravity and the mass moment of inertia of a ku-band antenna as closely as possible.

Test Site: All testing was conducted at the AFPEA, HQ AFLC/DSTZ, Building 70, Area C, Wright-Patterson AFB OH 45433-5999. The equipment required for each test is noted in the test plan.

TEST PROCEDURES AND RESULTS

1. Weight Test

Test Plan No. 1: The container was weighed to determine weight compliance.

Results: The cover weighed 230 pounds, the base weighed 506 pounds, total tare weight of 736 pounds. The results of this first are acceptable.

2. Leak Test

Test Plan No. 2: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2 at 1.00 psig. The vacuum retention test was conducted in accordance with FED-STD-101, Method 5009.2 at 1.00 psig. The failure criteria for the test was 0.0250 psig loss during a 30 minute period, after teperature and pressure stabilization.

Results: At the end of the 30 minute test period the pressure loss was 0.007 psig. At the end of the 30 minute test period the vacuum loss was .012 psig. The results of this test are acceptable.

3. Rough Handling Tests (+140°F)

a. <u>Test Plan No. 3a</u>: The high temperature cornerwise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5005.1. The height of the drop was 20 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 10 Gs was obtained during the test.

b. <u>Test Plan No. 3b</u>: The high temperature edgewise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5008.1. The height of the drop was 20 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 10 Gs was obtained during the test.

c. <u>Test Plan No. 3c</u>: The high temperature pendulum-impact test was conducted in accordance with FED-STD-101, Method 5012. The impact velocity was 7 ft/sec, the height of the drop was 9 inches (see figure 7).

Results: Visual inspection revealed no external damage to the container. A maximum of 19 Gs was obtained during the test.

The container was opened after the pendulum-impact test. Visual inspection revealed no damage to the container or the test load. The results of these tests are acceptable. See appendix 1 for detailed acceleration results.

4. Leak Test

Test Plan No. 4: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.004 psig. The results of this test are acceptable.

5. Rough Handling Tests (-20°F)

a. <u>Test Plan No. 5a</u>: The low temperature cornerwise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5005.1. The height of the drop was 20 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 20 Gs was obtained during the test.

b. <u>Test Plan No. 5b</u>: The low temperature edgewise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5008.1. The height of the drop was 20 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 20 Gs was obtained during the test.

c. <u>Test Plan No. 5c</u>: The low temperature pendulum-impact test was conducted in accordance with FED-STD-101, Method 5012. The impact velocity was 7 ft/sec, the height of the drop was 9 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 23 Gs was obtained during the test.

The container was opened after the pendulum-impact test. Visual inspection revealed no damage to the container or the test load. Upper cushion became unglued at the glue line. The results of this test are acceptable.

6. Leak Test

Test Plan No. 6: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.001 psig. The results of this test are acceptable.

7 Vibration Fatigue Test

Test Plan No. 7: The vibration fatigue test was conducted in accordance with MIL-STD-648, paragraph 5.3.2. The container was rigidly attached to the platform. A sinusoidal vibration excitation was applied in a vertical direction and cyclically expet or 7.5 minutes at 2 minutes per octave to locate the resenant frequency. Input from 5 to 12.5 Hz was at 0.125 inch

double amplitude and input from 12.5 to 50.0 Hz was at 1.0 G. A 30 minute dwell test was conducted at the resonant frequency.

Results: Visual inspection revealed no damage to the container or the test load. A maximum of 3.8 Gs was obtained at the resonant frequency of 7.7 Hz. The maximum transmissibility obtained was 5.0. The results of this test are acceptable.

8. Leak Test

Test Plan No. 8: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The results of this test are acceptable.

9. Repetitive Shock (Superimposed Load) Test

Test Plan No. 9: The repetitive shock test was conducted in accordance with MIL-STD-648, paragraph 5.2.2.1 and FED-STD-101, Method 5019.1. 926 pounds of steel plate was banded to the top of the container to simulate a stacked load. The container was not attached to the platform, but restraining blocks were attached to the platform to prevent the container from moving off the platform. The platform was vibrated at 3 to 5 Hz until the container raised from the platform (1/16 inch feeler gauge clearance between the container bottom and the platform), or a maximum platform acceleration of 1.0 G +/- .1 G was achieved. The test was run at the determined frequency for a period of two hours.

<u>Results</u>: Visual inspection revealed no damage to the container or the test load. A maximum of 10.5 Gs was obtained during the test. The results of this test are acceptable.

10. Leak Test

Test Plan No. 10: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.006 psig. The results of this test are acceptable.

11. Superimposed Load Test

Test Plan No. 11: The superimposed load test was conducted in accordance with FED-STD-101, Method 5016.1. A load of 5542 pounds was placed on top of the container through beams placed to simulate actual loading on the bottom container of a stack of four with a factor of safety of two.

Results: Visual inspection revealed no damage to the container. The results of this test are acceptable.

12. Leak Test

Test Plan No. 12: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The results of this test are acceptable.

13. Stand Off Test

Test Plan No. 13: The stand off test was conducted in accordance with paragraph 4.7.5.1 of MIL-C-5584D. The cover was set on a concrete surface with the stand offs in contact with the floor. A load of 465 pounds was then placed on top of the cover. With the load removed, the cover was then slid on the stand offs five feet in each of four directions.

Results: Visual inspection revealed no deflection or deformation with the 465 pound load on top. No physical damage resulting in a loss of functional performance was found following the sliding. The results of this test are acceptable.

14. Mechanical Handling Tests

Test Plan No. 14a: The forklift handling test was conducted in accordance with FED-STD-101, Method 5011.1, paragraph 6.2 (see figure 8). 1x4 inch boards were used since the forklift used has hard rubber tires. Stack one container on another container and varify forklift entry from all four sides.

Results: During the test the container was stable riding on the times. Visual inspection revealed no damage to the times. Forklift entry was possible on all four sides. The test of this test are acceptable.

The Plan No. 14b: The forklift pushing test was conducted with FED-STD-101, Method 5011.1, paragraph 6.5.

<u>Results</u>: Visual inspection revealed no functional damage to the container. The results of this test are acceptable.

c. <u>Test Plan No. 14c</u>: The forklift towing test was conducted in accordance with FED-STD-101, Method 5011.1, paragraph 6.6.

Results: Visual inspection revealed no functional damage to the container. The results of this test are acceptable.

15. Hoisting Strength Test

a. <u>Test Plan No. 15a</u>: The single ring hoisting test was conducted in accordance with MIL-STD-648, paragraph 5.8.5. The loaded container was lifted by a lift ring and suspended for five minutes.

Results: Visual inspection of the container revealed no damage or deformation. The results of this test are acceptable.

b. Test Plan No. 15b: The 4 ring hoisting strength test was conducted in accordance with MIL-STD-648, paragraph 5.8.3. The container was loaded with 4030 additional pounds and hoisted by all four lift points on the container base simultaneously and left hanging for five minutes (see figure 9).

Results: Visual inspection revealed no damage to the container. The results of this test are acceptable.

c. <u>Test Plan No. 15c</u>: The 4 ring hoi ting strength test was conducted in accordance with a modified version of MIL-STD-648, paragraph 5.8.3. This test was conducted because the container cover also has lift rings, however it requires only the weight of one container instead of five. The container cover was loaded with 712 additional pounds and hoisted by all four lift points simultaneously and left hanging for five minutes.

Results: Visual inspection revealed no damage to the container. The results of this test are acceptable.

16. Leak Test

Test Plan No. 16: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.004 psig. The results of this test are acceptable.

17. Cover Handle Pull Test

Test Plan No. 17: The cover handle pull test was conducted in accordance with paragraph 4.7.4.1 of MIL-C-5584D. One handle was used to lift the 230 pound cover off the ground. A 20 pound weight was placed in the cover to give a total weight of 250 pounds. The cover was maintained in that position for 5 minutes (see figure 10).

Results: Visual inspection revealed no deflection or permanent deformation to the cover handle or the container cover. The results of this test are acceptable.

18. Puncture Resistance Test

Test Plan No. 18: The pendulum puncture test was conducted in accordance with MIL-STD-1489A, Method 505. The test apparatus used was a simulated forklift tine weighing 70 pounds suspended by wire cables. The tine was pulled straight back until it reached a height of 28 inches above its equilibrium height and released. The tine impact was made to each side and end of the unrestrained container at one inch above the forklift pocket.

Results: Visual inspection revealed no functional damage to the container. The results of this test are acceptable.

19. Leak Test

Test Plan No. 19: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The results of this test are acceptable.

20 Towing Test

Test Plan No. 20: The towing test was conducted in accordance with FED-STD-101, Method 5011.2. The test apparatus used was a mobility container towbar (part number 305435-1). This was attached to the towbar attachment points on the container. The container was towed for 22 feet in one minute and was thanked twice at 180°. The container shall not sway more than two first in any direction.

Results: The container rolled smoothly without excessive arong when towed. Visual inspection revealed no damage to the results of this test are acceptable.

21. Structural Integrity Test

Test Plan No. 21: The structural integrity test was conducted in accordance with MIL-STD-648, paragraphs 5.5.2 and 5.5.3. The pressure relief valve was removed and the loaded container was sealed. The container was pressurized to 3.5 psig. The pressure was not maintained for any length of time and leakage was not monitored. The test was conducted to determine if permanent structural deformation, deformation that prevented removal of the contents, or any potentially unsafe conditions occurred. Similarly, the container was subjected to a vacuum of 2.5 psig and observed for the same failure criteria as above.

Results: Visual inspection revealed no damage or permanent deformation to the container. The results of this test are acceptable.

CONCLUSIONS

The CNU-459/E prototype container provided protection for the contents when tested in accordance with the container test plan.

RECOMMENDATIONS

Safe for worldwide shipment and storage of one ku-band antenna.

				Table 1.	Test Pl	an			
	AIR FORC	_			TION AC	TIVITY	,	AFPEA PRO	JECT NUMBER
		((Container T	est Plan)				88-P-1	.03
C	ONTAINER SIZE	(LxWx)	D) (INCHES)	WEIGH	(LBS)	CUBE (C	J. FT.)	QUANTITY	DATE
	INTERIOR:	1	TERIOR:	GROSS:	ITEM;				05.35
TEM I	100 A 00 E	64 X 6	2 x 52	942	206	119		1	26 May 8
					MANUI	FACTURER -	_		•
	nbat Talon	II Ku-	band Ant	enna				on Elect	
							CO	NTAINER CO	ST
	J-459/E DESCRIPTION								turne de la company de la comp
	minum Cont	ainer							
	noted belo		i			ſ			ı
NO.	REF STD/S AND TEST ME PROCEDURI	THOD OR	TEST TI	TLE AND PA	RAMETERS			TAINER ITATION	INSTRU- MENTATION
1.	WEIGHT TH					1			
	(1)[4.7.1	LOJ		er cover				ssem-	Scale
			than 240	ot be gre	eater Total		ed co cludi	ntainer	
			,	er weigh				.ng .sola-	
	:		not be	greater '				stem.	
	:		750 lbs.	•					
2.	LEAK TEST	,							1
۷.	FED-STD-1		Pneumati	c press	re at	Tes	t pe	rformed	Water
	Method 50		1.00 PS	and va	cuum	in	ambi	ent	manometer
	[4.7.2]		retention			1		on from	
	:		.025 PSI	leakage				sed air vacuum	
			stabilia			pun		Vacuum	
			duration		a minimu		•		
	;		of 30 mi	nutes.					
;.	ROUGH HAN	DLING	TESTS (HI	GH TEMPI	ERATURE	+140°E	')		
	FED-STD-1		Cornerwi					p on	Tri-axial
	Method 50	05.1	tional)			on dia	gona	ī	accelero-
	7.83	:	at +140°					cor-	meters
			than 24 height 2					otal of ps.(2)	
;	1			2	,,,,			PO: (2)	j
••	?-STD-1		Edgewise					p on	Tri-axial
	:Method 50 -14.7.31	08.1	tional) at +140°					acent edges,	accelero- meters
	* * * * * 5 •	1	than 24		Drop			euges, f two	meters
		į	height 2				ps.(
		;	•				•		ļ
		!							
		i							j
5 - 4	1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1		brackets						
*	A		ormer dr age drop						
	6) Xa		ant acce	leration	shall	not ex	ceed	-25 G's	
Diể≎ (.	يمن شي		I. L. MAN/R		,		/ 42/		
∂./ ⊝]	ine Bucke	y, Mech	anical E	ndneer	Ted Hi	nds, C	hief	, Design	Br., AFPEA

	AIR FORCE				ON AC	HIVITY		ROJECT NUMBER
				Test Plan)			88-P-	
C	ONTAINER SIZE (I INTERIOR:	EX	TERIOR:	WEIGHT (GROSS:	ITEM:	1	T.) QUANTITY	
PEM I	NAME	64 × 6	2 x 52	942	206	119	. 1	26 May 8
		TT V	h 3		MARUF	ACTURER	erson Elec	
	abat Talon :	II Nu-	Dand And	enna	. 	Eme	CONTAINER C	· · · · · · · · · · · · · · · · · · ·
	J-459/E						CONTAINEN	,031
	DESCRIPTION						1	
Alu	minum Conta	iner						
	FIONING							
As	noted below	1.						
EST NO.	REF STD/SP AND TEST MET PROCEDURE	HOD OR	TEST T	ITLE AND PARA	METERS		CONTAINER RIENTATION	INSTRU- MENTATION
c.	FED-STD-10	1	Pendulu	m-impact t	test.	One :	impact on	Tri-axial
	Method 501	.2		on at +165		1	side and	accelero-
	[4.7.8]		. •	ture of shion system		one of tw	end, tota:	l meters, Thermo-
		ļ		test shall			cts.(3)	couples
				$(+10/-0^{\circ}F)$		•	, ,	
				velocity 7				
			inches.	<pre>drop heig (4)</pre>	int 9			
4.	LEAK TEST	; !						
į	FED-STD-10			ic pressur			ent	Water
	Method 500 [4.7.2]	9.2		I. Test d s than 15				manometer
	[4.,.2]			125 PSI le		-5		
			allowed	•	-			i i
5.	ROUGH HAND	LING	TESTS (L	OW TEMPERA	TURE -	-20 ^(F)		! :
a.	FED-STD-10			ise-drop (lrop on	Tri-axial
;	Method 500	5.1	tional)	test. Co	nditio			accelero-
!	[4.7.8]	!		F for not hours. D	rop		om cor- total of	meters
:				20 inches.	-		rops.(5)	
h	FED-STD-10	,	Edgowie	e-drop (ro	+	One d	lrop on	Tri-axial
D .	Method 500			test. Co			ottom	accelero-
ļ	[4.7.8]	-	at -20 ⁶	F for not	less	edges	, total	meters
! !		ļ		hours. D 20 inches.	•		o drops.	
			nergiic	zo inches.	(4)	(6)		
MME	NTS:							
,	5) These co	orners	are on	ocite the	co imn	acted in	Test No.	79

	AIR FORCE	PACKAGING I		ON A	CTIVIT	TY AFPEA PROJECT NUMBER		
		(Container 1	'est Plan)				88-P-1	03
C	NTAINER SIZE (L.	x W x D) (INCHES) EXTERIOR:	WEIGHT	LBS) ITEM:	CUBE (CU. FT.)	QUANTITY	DATE
Tem N	64	x 62 x 52	942	206	119		1	26 May 8
		T Vu-band lat	0000		, acione		on Elect	ric
	INER NAME	Ku-band Ant	enna				NTAINER CO	
••••						CO	MIAINER CO	91
	-459/E							
	DESCRIPTION							
	minum Contai	ner						
ONDIT	TIONING							
As	noted below.							. p. n. n. n. man an menan manakan menan
EST NO.	REF STD/SPEC AND TEST METHO PROCEDURE NO	DO OR TEST TO	TLE AND PARA	METERS			TAINER ITATION	INSTRU- MENTATION
c.	FED-STD-101	Pendulu	m-impact t	test.	Or	ne imp	act on	Tri-axial
	Method 5012	Conditi	on at -65°	PF.	or	ne sid	le and	accelero-
	[4.7.8]		ture of sl				l, total	meters,
			ion syster test shal		1	two	(6)	Thermo- couples
			+0/-10 ^O F)		1.1	paces	. (0)	Couples
			velocity					1
			drop heig					
		inches.	(4)					
ڻ. ;	LEAK TEST FED-STD-101	Pneumat	ic pressur	a wit	-h An	bient	•	Water
	Method 5009		I. Test			DICHE	•	manometer
	[1.7.2]		s than 15					
			125 PSI le	eakage	≥			
		allowed	•		Ì			
:	STATION F	ATIGUE TEST						
	MIL-STD-648		xcitation	of	Ri	gidly	attach	Tri-axial
i	Para 5.3.2		nch double				er to	accelero-
!	[4.7.7.1]		de or 1G,			citer		meters,
		i i	er is less				straps	Thermo-
			proximate			pron	ibited.	couples
			nmically f z (about 1		•			
,			nin) for 7		1			
		minutes	-		80			
,		· ·	at the re					
			y. The t		ay			
			rupted to					
		prevent	excessive	:				
		1						1
		:						
		•						

APPROVED BY: PAREATE, TY. Carolina Buckey, Mechanical Engineer Ted Hinds, Chief, Design Br., AFPEA

C	ONTAINER SIZE (L x W x INTERIOR: E)	D) (INCHES) (TERIOR:	WEIGHT GROSS:	(LBS)	CUBE (CU. F	T.) QUANTITY	DATE
	64 x 6	52 x 52	942	206	119	1	26 May 8
TEM !				MANUF	ACTURER		
	nbat Talon II Ku- AINER NAME	-band Ant	enna			rson Elec	
	J-459/E						
	DESCRIPTION						
	minum Container			٠		,	-
As	noted below.				.,		,
NO.	REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST T	ITLE AND PAR	AMETERS	1 -	ONTAINER IENTATION	INSTRU- MENTATION
		materia sibilit	ture rise ls. Tran y shall n 5 at the cy.	smis- ot	nt	•	
3.	LEAK TEST FED-STD-101 Method 5009.2 [4.7.2]	1.00 PS	ic pressu I. Test o s than 15 125 PSI lo	duratio minuto	on	nt	Water manometer
•	REPETITIVE SHOCK MIL-STD-648 Para 5.2.2.1 FED-STD-101 Method 5019.1 [4.7.7.3]	Test us motion 1 G or between (whicher Banding shall be forklift the char	ing vertice for two helds to 5 Helds of contact through the contact the contac	cal ours and ours and ours the and incomes		nt	Tri-axial accelero- meters
0.	LEAK TEST FED-STD-101 Method 5009.2 [4.7.2]	1.00 PSI not less	ic pressur . Test o s than 15 .25 PSI le	duration minute	n	nt	Water manometer
OMME REPAF	NTS:			APPROVEC	BY:		

	AIR FORC	_			ION AC	TIVITY	AFFEA PRO	DJECT NUMBER
		((Container T	est Plan)			88-P-	103
C	ONTAINER SIZE		D) (INCHES) TERIOR:	WEIGHT GROSS:	(LBS)	CUBE (CU. FT.)	QUANTITY	DATE
		· ·	2 x 52	942	206	119	1	26 May 8
TEM I	NAME				MANUF	ACTURER	•	the second secon
	mbat Talon	II Ku-	band Ant	enna		Emer	son Elec	tric
ONT	INER NAME					C	ONTAINER CO)ST
	J-459/E			- · · ·	*******			
ACK	DESCRIPTION							
	minum Cont	ainer						to the second second second in the second
ONDI	TIONING							
As	noted belo							g to the contract of the contract of
rest No.	REF STD/S AND TEST ME PROCEDURE	THOD OR	TEST TI	TLE AND PAR	AMETERS		TAINER NTATION	INSTRU- MENTATION
	· · · · · · · · · · · · · · · · · · ·		* **			-		
1.	SUPERIMPO		AD					
	FED-STD-1		t .	containe				Record
	Method 50			whichev		contai		changes,
	4.7.6.1			Load e		contai		i.e.
				bottom c			test.	buckling,
				factor o				deforma-
				eave stac ces at am		·		tions.
			temperat		prenc			
_			•			}		
3.	LEAK TEST		, 					
	FED-STD-1 Rethod 50			c pressu . Test			:	Water
	La2]	09.2		than 15				manometer
	[4,,,,5]			.25 PSI 1		; 5		
			allowed.		eanage			
•	CELVE OFF	mnam !						
♪•	STAND-OFF [1.7.5.1]	TEST	Diana la			e Place o		724 3
	11.7.5.1			ight on		, , , , , , , ,		Visual
		1		r shall		tainer on a co		inspection
				r deflec				
				oved, sl		on the		
				the star				
				t in eacl				
			four dif	ferent d:	irec-			
		i		There sha				
		į		e to the	sealin	g		
		į	surface.			İ		
								1
		;						1
	•	•				1		1

APPROVED BY:

AFALL No. 1.4 PAGE 5 OF 1

, PAERAR 10 93.

AIN FUNC		KAGING Container 1	EVALUAT (est Plan)	ION AC	TIVITY		ROJECT NUMBER
CONTAINER SIZE	(L x W x		magnini anno magnini	(LBS)	CUBE (CU. F	88-P T.) QUANTIT	
MIENION:	1	52 x 52	942	206	119	1	26 May 8
M NAME			<u></u>	MANUF	ACTURER		
Combat Talon	II Ku-	band Ant	enna		Eme	erson Ele	ctric
NTAINER NAME						CONTAINER	COST
NU-459/E	.,44- 4						
K DESCRIPTION							
luminum Cont IDITIONING	. •						
s noted belo		·					
REF STD/S AND TEST ME PROCEDURE	THOD OR	TEST T	ITLE AND PAR	AMETERS		CONTAINER	INSTRU- MENTATION
. HANDLING FED-STD-1 Method 50 Paragraph [4.7.5]	01 11.1	Lift conground clined stack reprevent 100 ft Place to 54 inche path as from the to the state angle so wheel so 90 ft fra 75° and the tine test. The structure stack or top of and verientry called the stack of the structure stack or top of and verientry called the stack of the structure stack or top of and verientry called the stack or top of an and verientry called the stack or top of an another top of an anot	t handling ntainer of with time of the content of t	off es in- es and i to Carry conds. lel 2x4s in the 30 ft square ft from the lef carry	y s n ft at s s s s s s s s s s s s s s s s s s	ent	Visual inspection

AIR FORC		- • · · · - - ·		ION AC	TIVITY	AFPEA PROJECT NUMBER			
		Container 1	rest Plan)			1	88-P-103		
CONTAINER SIZE INTERIOR:	EX.	O)(INCHES) TERIOR: 2 x 52	WEIGHT GROSS: 942	(LBS) ITEM: 206	CUBE (CU. 1	T.) QUANTI	DATE 26 May 8		
TEM NAME	104 X 0		342		ACTURER		20 May 6		
Combat Talon	TT Ku-	band Ant	enna			erson El	ectric		
CONTAINER NAME						CONTAINER			
CNU-459/E									
PACK DESCRIPTION						L			
Aluminum Con	tainer								
CONDITIONING									
As noted bel	ow.								
TEST REF STD/S	ETHOD OR	TEST T	ITLE AND PAR	AMETERS		CONTAINER RIENTATION	INSTRU- MENTATION		
b. FED-STD- Method 5 Paragrap [4.7.5]	101 011.1	The for should not sur tainer. surface seconds	t pushing klift tinextend unport the Push on 35 ft in the cutting	nes nder bu con- n hard n 85 shall	and cont	one sid one end ainer.			
c. FED-STD- Method 56 Paragraph	011.1 h 6.6	Pull by ft in 2 shall be damage.	t towing towing towing towing towing towing towing towing the seconds of the second of the	rings 1 . The	00 cont	ends of ainer	Visual inspection		
15. HOISTING			ring hois	sting	Ambi	ant	Visual		
(4.7.4]		test. from ea turn an five mi shall b	Hoist con ch lift p d suspend	ntainer point in d for There age or		enc	inspection		
8. WIL-STD-0 20 va 5.8 3.7.4)		Hoist of to five weight contain points	ng hoisti ontainer times the of a since er by all simultane ve hangir . There	loaded ne gross gle lift cously	5	ent	Visual inspection		
, e e				•			1		
Fallet Lander				AFPROVE) BY:	and the second section is a			

CAROLL & ELCKe, Mechanical E gineer Ted Hinds, Chief, Design Br., AFPEA

		(AGING EVALUATION ACT	TIVITY	AFPEA PRO	JECT NUMBER
C		TERIOR: GROSS: ITEM:	CUBE (CU. FT.)	QUANTITY	DATE
ITEM I		2 x 52 942 206 MANUFA	119 CTURER	1	26 May 89
	nbat Talon II Ku-	band Antenna		on Elect	
	INER NAME		Co	NTAINER CO	ST
	DESCRIPTION				
	minum Container		. Lagrang scales a 1 communication		
	noted below.				
TEST NO.	REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST TITLE AND PARAMETERS		TAINER ITATION	INSTRU- MENTATION
		no damage or permanent deformation. (Legs of the sling should be 30° from the horizontal). Four ring hoisting test for cover lift rings. Hoist container loaded to one times the gross weight of a single container by all lift points simultaneously and leave hanging for 5 minutes. There shall be no damage or permanent deformation. (Legs of the sling should be 30° from the horizontal).	Ambient	:	Visual inspection
16.	LEAK TEST FED-STD-101 Method 5009.2 [4.7.2]	Pneumatic pressure with 1.00 PSI. Test duration not less than 15 minutes with .0125 PSI leakage allowed.			Water manometer
COMME					
	RED BY:	APPROVED		Docies	D~ APPDA
	FORM 4	nanical Engineer Ted Him	ius, chiel	, pesidu	PAGE 8 OF 11

	AIR FORCE PAC			ON AC	TIVIT	Y	AFFER PROJ	ECT NUMBER
L		Container T					88-P-10	
C	ONTAINER SIZE 'L x W x	D) (INCHES) TERIOR:	WEIGHT	(LBS) ITEM:	CUBE (CU. FT.)	QUANTITY	DATE
_	64 x 6	2 x 52	942		•		. 1	26 May 89
ITEM N				MANUI	FACTURE		nam Black:	-i -
	nbat Talon II Ku- NMER NAME	band Anto	enna				ONTAINER COS	
	1-459/E DESCRIPTION							·
	minum Container MONING				•			
As	nated below.					** ***		
TEST NO.	REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST TI	TLE AND PARA	METERS			TAINER ITATION	INSTRU- MENTATION
17.	COVER HANDLE PU							
	[4.7.4.1]		force of ver handle		bs A	nbient	t .	Scale
			rections					
			loads are		Ì			
		be no da	e. There	Snall				
			nt deforma	ation.				
18.	PUNCTURE RESIST.	ANCE TEST	<u>c</u>					
	MIL-STD-1489A		vill be ma				pact to	Visual
	Method 505.1		/2 inch al 1 forklif				de and d of the	inspection
		of the d	container	base.	C	ontair	ner	
			nall not laffecting			•	cotal of cacts.	
	•		performan			40 Ting	paoco.	
	ED-STD-101	Proumati	ic pressui	ro wit	h A	nbient	•	Water
,		1.00 PSI				IDTEIL		manometer
	[4.7.2]	not less	than 15	minut	es			
		with .01 allowed.	25 PSI 1	eakage				
		allowed.						
					1			
	i							
	,						:	
	į							
	!							
	;							
	!							
		,						
等 名・27年	- - 변경 왕조		12	APPROVE	D BY:			
isro)	Buckey, Mech	anical E	gineer	Ted H	inds,	Chief	, Design	Br., AFPEA
	10 V 21 4			8				PAGE 9 OF 1

	AIR FORCE	AFPEA PROJECT NU 88-P-103					
С	ONTAINER SIZE (L INTERIOR:	EXTERIOR:	WEIGHT (ITEM:	•	T.) QUANTITY	DATE
ITEM		4 x 62 x 52	942	206	119 ACTURER	1	26 May 89
	nbat Talon II	[Kushand An	tanna	MARUF		erson Elec	atria
	LINER NAME	L Nu Dalla All	CCIIIIG		EME	CONTAINER C	
CNU	I-459/E						
	DESCRIPTION						
	minum Contai	ner					
	TIONING						
As	noted below.						- ,
TEST NO.	REF STD/SPEC AND TEST METHO PROCEDURE NO	DD OR TEST	TEST TITLE AND PARAMETERS		1	ONTAINER RIENTATION	INSTRU- MENTATION
20.	TOWING TEST			.99 %-	2-1-2		77 i mu - 3
	FED-STD-101 Method 5011	1	ntainer sha behind a fo			ent	Visual Inspection
	1.001104 3011		a mobility	* \T T T (-		Tuebeccion
		contair	ner towbar				•
		•	5-1) attach				·
			bar attach on the con		- :		
			container				
		least 2	00 ft. in	1_			
; •			turning 18				1
‡ 1			wice durin. There sh				
			ige to the				
•		attachm	ent points	and			
ŧ			tainer sha				i
1			ore than 2	ft. in	l		
:		any air	ection.		 		ì
į		1 :					
;		:					
t							
					}		
1							
İ							
-							
					1		
Ī							
ļ					}		
1							
) لماريونو						w	.1
OMME	NTS:						
REPAR	EO BY:		A	PPROVED	BY:		a design of the section of the secti
arol	ine Buckey,	Mechanical	Engineer '	Ted Hi	nds, Chi	ef, Desig	n Br., AFPEA

AIR FORC			EVALUATION PORTION PROPERTY PR	ON AC	TIVITY			ECT NUMBER
	(6	ontainer 1	Test Plan)				88-P-1	.03
CONTAINER SIZE INTERIOR:	EXT	ERIOR:	GROSS:	ITEM:	CUBE (CU. I	T.)	ļ	
	64 x 6	2 x 52	942	206	119		1	26 May 89
ITEM NAME				MANUF	ACTURER			_
Combat Talon	II Ku-l	oand Ant	tenna		Em	ers	on Elect	ric
CONTAINER NAME						CO	NTAINER COS	T
CNU-459/E								
PACK DESCRIPTION						i		The Control of the Co
Aluminum Con	tainer							
CONDITIONING								
As noted belo	ow.							
TEST REF STD/S AND TEST ME PROCEDURI	ETHOD OR	TEST T	TITLE AND PARA	METERS			TAINER	INSTRU- MENTATION
MIL-STD-OPARAGRAPH 9.5.2. & [4.2.2.1	648 hs 5.5.3 .16]	Pressur 3.5 PSI return a vacuumomenta to ambi resulti an unsa shall k rejecti deforma contair suffici the sai antenna	rize contain momentaring to ambient in of 2.5 For in or carrier and respond to the cause for carrier to present to presen	App SI return ailur ausin on or le le le le le le le le le le le le le	d ly e g	ent		Water manometer
	į							

APPROVED BY:

APPROVED BY:

Ted Hinds, Chief, Design Br., AFPEA

AFA - MOVE 4

20 PAGE 11 OF 11

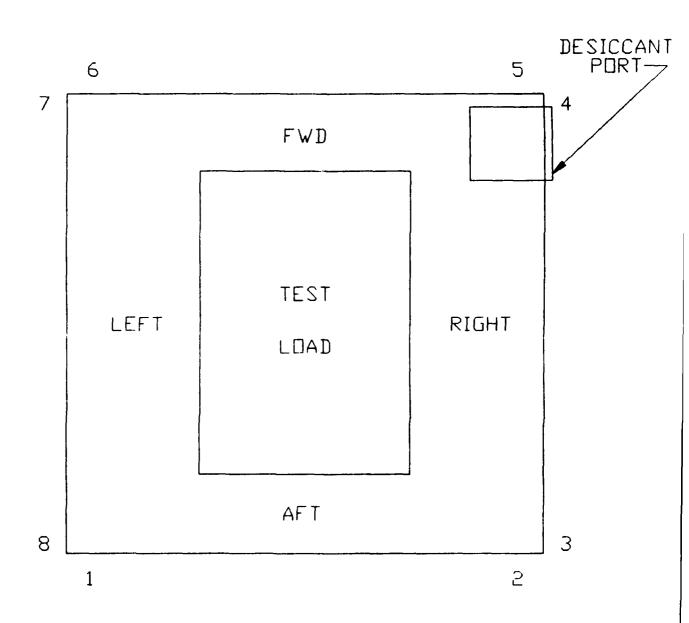


FIGURE 1 - CNU-459/E Corner Numbering

Figure 2

CNU-459/E Prototype Container.

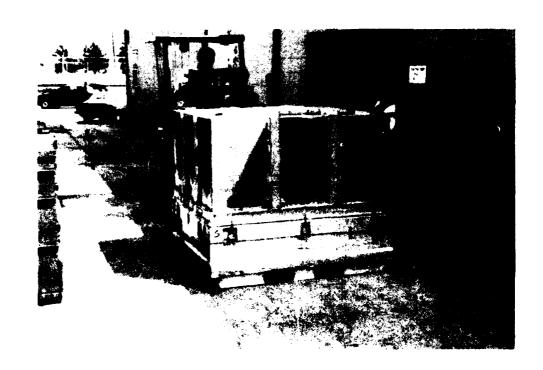


Figure 3

CNU-459/E Yoke Assembly Attachment.

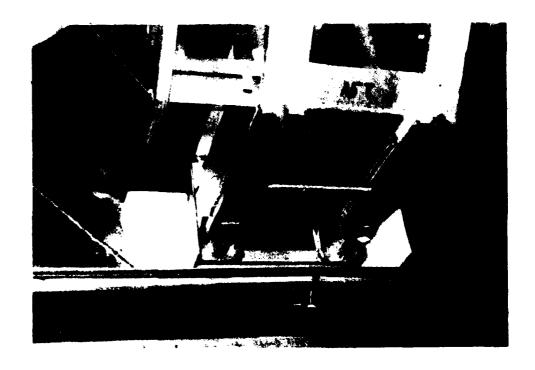


Figure 4

CNU-459/E Cover Wheel Storage.

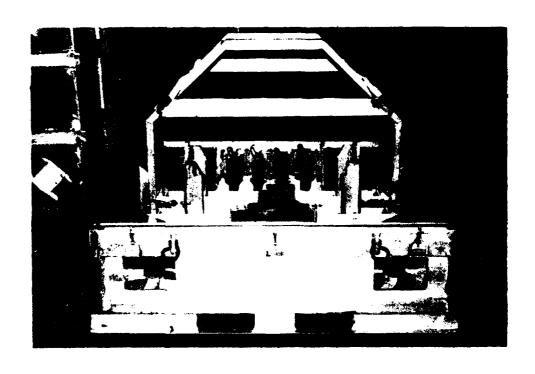


Figure 5

CNU-459/E Yoke Frame.

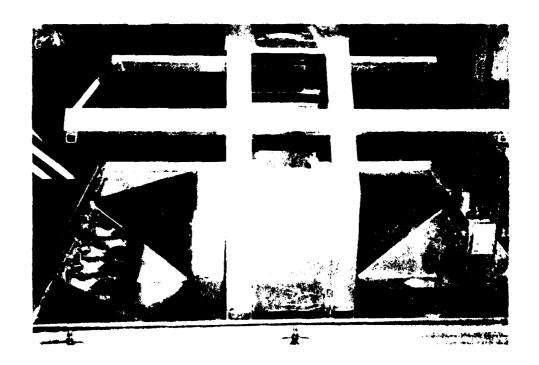


Figure 6
CNU-459/E
Test Load.



Figure 7

High Temperature Pendulum-Impact Test.



Figure 8

Forklift Handling Test.



Figure 9

Four Ring Hoisting Test.

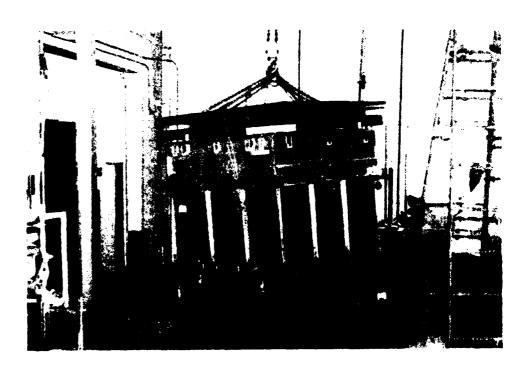
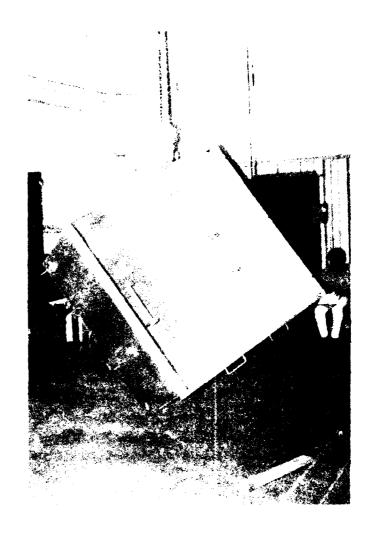


Figure 10
Cover Handle Pull Test.



DISTRIBUTION LIST

DTIC/FDAC Cameron Station Alexandria, VA 22304-6145	2
HQ AFLC/DSTZ Library Wright-Patterson AFB OH 45433-5999	10
HQ AFLC/DSTTP Wright-Patterson AFB OH 45433-5999	1
HQ USAF/LETTC Washington DC 20330	1
HQ AFSC/LGT Andrews AFB DC 20334-5000	1
OC-ALC/DSTD Tinker AFB OK 73145	1
OO-ALC/DSTD Hill AFB UT 84406	1
SA-ALC/DSTD Kelly AFB TX 78241	1
SM-ALC/DSTD McClellan AFB CA 95652	1
WR-ALC/DSTD Robins AFB GA 31098	1
WR-ALC/DST Robins AFB GA 31098	1
OO-ALC/MMIH/MMWR Hill AFB UT 84406	2
ASD/ALXP/SDM Wright-Patterson AFB OH 45433	1
GSA, Office of Engineering Mgt Packaging Division Washington DC 20406	1
AD/MSD/YBAC	1

DISTRIBUTION LIST (Cont'd)

Commander Naval Supply Systems Command Attn: N. Karl (SUP 0611F) Washington DC 20376-5000	1
Commander Naval Air Systems Command Attn: E. Panigot (AIR 41212A) Washington DC 20361	1
Commander Space and Naval Warfare Systems Command Attn: T. Corbe (Code 8218) Washington DC 20360	1
Commander Naval Facilities Engineering Command Hoffman Bldg. #2, Room 12S21 Attn: C. Manwarring (FAC 0644) Alexandria, VA 22332	1
Commanding Officer Naval Construction Battalion Center Attn: K. Pollock (Code 15611K) Port Hueneme, CA 93043	1
Commander Naval Sea Systems Command Attn: F. Basford (SEA 05M3) Washington DC 20362	1
Commanding Officer Naval Aviation Supply Office 700 Robbins Avenue Attn: H. FURLONG Philadelphia, PA 19111-5098	1
Commanding Officer Navy Ships Parts Control Center P.O. Box 2020 Attn: F. Sechrist (Code 0541) Mechanicsburg, PA 17055-0788	1
Commanding Officer Naval Air Engineering Center Attn: F. Magnifico (SESD Code 9321) Lakehurst, NJ 08733-5100	1

DISTRIBUTION LIST (Cont'd)

Commanding Officer Naval Weapons Station, Earle ATTN: NWHC 80A (Mel Gray) NWHC/Code 8023 Colts Neck,NJ 07722-5000	1
US AMC Packaging, Storage, and Containerization Center/SDSTO-T Tobyhanna, PA 18466-5097	1
DLSIE/AMXMC-D US Army Logistics Mgt Ctr Ft Lee VA 23801-6034	1
US Army ARDEC/SMCAR-AEP Dover, NJ 07801-5001	1
HQ DLA/OWO Cameron Station Alexandria, VA 22304-6145	1
ASD/VXAL Wright-Patterson AFB OH 45433	2
ASD/VXA Wright-Patterson AFB OH 45433	1

CNU-459/E CONTAINER - DETAILED ACCELERATION RESULTS

HIGH TEMPERATURE ROUGH HANDLING TESTS (+140°F)

			readings (Gs)	
Impact	Position	Right	CG	Left
20" rotational drop	Corner 2-3	9	8	8
20" rotational drop	Corner 6-7	7	7	10
20" rotational drop	Side 5-6	7	N/A	7
20" rotational drop	Side 7-8	8	7	10
7 ft/sec pendulum-impact	Side 1-2	12	4	7
7 ft/sec pendulum-impact	Side 3-4	19	7	11

1. No damage to the container or the test load.

LOW TEMPERATURE ROUGH HANDLING TESTS (-20°F)

Impact	Position	Accelerometer Right	readir CG	ngs (Gs) Left
20" rotational drop	Corner 2-	- -	12	14
20" rotational drop	Corner 6-		14	14
20" rotational drop	Side 5-6		16	13
20" rotational drop	Side 7-8		11	13
7 ft/sec pendulum-impact	Side 1-2	18	11	23
7 ft/sec pendulum-impact	Side 3-4	17	8	16

1. No damage to the container or the test load.

VIBRATION FATIGUE TEST

Natural frequency 7.7 Hz

(input: 0.38 G peak, 0.125 inch double amplitude)

	Lower	Upper	Left
	CG	CG	Dish
Maximum Acceleration (Gs, peak to peak) Maximum Transmissibility	3.7	3.8	3.6
	4.9	5.0	4.7

1. No damage to the container or the test load.